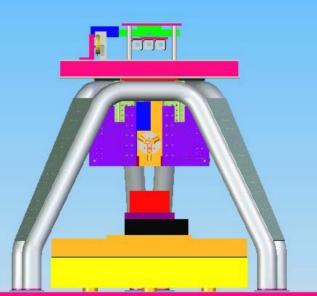
Laser Microengineering and the Advances that can be Gained by Using the Jefferson Laboratory Free Electron Laser

Henry Helvajian The Aerospace Corporation Los Angeles, California

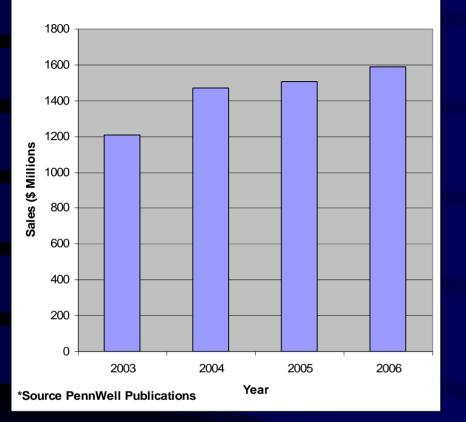
The Laser Microengineering Experimental Station" at the Jefferson Laboratory Free Electron Laser Facility



Enable Development of Laser Materials Processing Technology in the USA

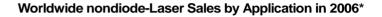


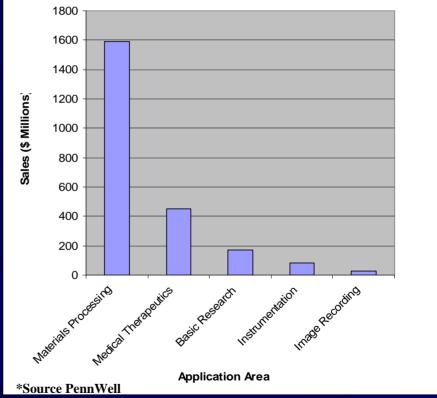
Worldwide nondiode-Laser Sales for Materials Processing Applications\*



#### ~ 1.6 \$B in Laser Sales Alone Expected in 2006

#### Materials Processing Applications Lead World Wide Sales





**Publications** 



## **Laser Microengineering**

Controlled Alteration of a Material Property in the Surface or Volume that with Patterning Leads to the Development of a Structure/Device/Component

#### **Examples of Processes**

- Single laser volumetric exposure in photoactive materials (e.g. holography).
- Multi laser volumetric exposure in active materials (e.g. photonic crystals).
- Percussion and ablative machining
- Polishing (e.g. mirrors coatings)
- Laser chemical vapor deposition (LCVD)
- Laser induced plasma deposition
- Laser induced forward transfer (LIFT)
- Laser induced phase transformation (i.e. crystallization, amorphous )
- Fusing (i.e. welding on the micron scale)
- Bending (i.e. curvature control on the nanometer scale)
- Texturing (i.e. control of surface topology)
- Pulsed laser desorption (atoms/molecules) and ablation (monolayers or greater).



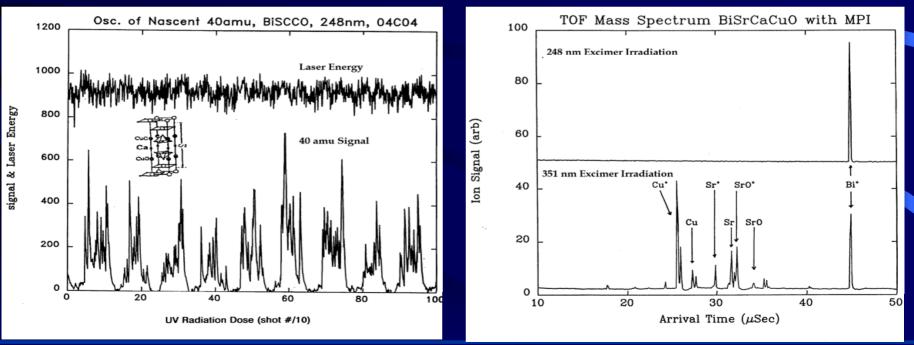
## "Rules of thumb" for Laser Microengineering

- In inducing a physical process by laser irradiation, there is more control with multiple small-energy laser pulses rather than a single large-energy pulse.
- Laser material interaction processes that are mostly driven by non-thermal events are likely to be more precise than processes governed by thermal induced phenomena.
- Tuning to a laser material interaction resonance helps.



#### **Examples From Prior Experiments** "atomic layer-by-layer pealing & species specific desorption"\*

- Low-KE Pulsed Atomic Beam Source for Controlling Epitaxial Deposition.
  - Practical applications
    - Atomic beam of specific species "layer-by-layer" atomic pealing of surfaces.
  - Impracticality
    - Measured removal rates  $10^{-7} 10^{-2}$  monolayers/shot ( $10^{11}$  species/cm<sup>2</sup> per shot).
  - Scaling
    - At MHz laser repetition rate deposition rates can be made to compete with MBE (deposition rates 0.1ML/s<sup>-1</sup>).



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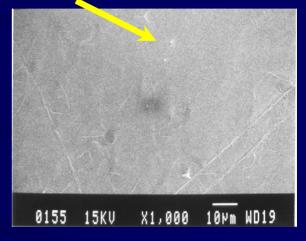
\* H. Helvajian, L. Wiedeman and H. –S. Kim "Photophysical Processes in Low-fluence UV Laser-Material Interaction and the Relevance to Atomic Layer Processing", Adv. Mat. For Opt. And Elect. Vol. 2 (1993) 31-42.



#### **Examples From Prior Experiments** "surface texturing in the non-thermal domain"\*

- Laser induced surface texturing in the "non thermal domain".
  - Practical Applications
    - Fabrication of a catalytically active surface on metals.
    - Nanometers scale control surface topological features
  - Scaling
    - 1 KHz laser can prepare surface at speeds ~ 0.005 mm/sec
    - 1 MHz laser can prepare surface at speeds ~ 5mm/sec.

15K Laser Shots (355nm) Low Fluence (<30mJ/cm<sup>2</sup>)





Henry.Helvajian@aero.org Department of MicroNanotechnology <sup>\*</sup>D. P. Taylor and H. Helvajian "Pulsed UV laser induced desorption of ions from Aluminum", Surface Science 451 (2000) 68-75.



- Practicalities of Industry
  - Utilizing large number of laser pulses per unit area means increased processing time and therefore increased cost.
    - Advantages of processing material in the "non-thermal regime" are nullified because of the large number of required laser pulses. These processes become relegated as scientific observation and deemed not practical.
- Possible "game" or paradigm changers
  - A laser with high repetition rate (>MHz) and high average power that is wavelength tunable.
  - A means for controllably delivering, with high fidelity, laser pulses with prescribed amplitudes (energy) to a material that is moving under pattern control.
  - A laser material process in which the laser light only "activates" the material but the desired physical transformation occurs in a follow-on batch process (e.g. chemical etch step)



**The JLAB IRFEL & UVFEL** 

Unique Light Sources for Laser Material Processing

**High Repetition Rate - Tunability** 



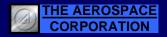
The Laser Microengineering Experimental Station (LMES) at JLAB

A Three Axes Motion Tool Designed For the Controlled Delivery of the FEL Light to a Moving Surface (speeds > 400mm/sec) with Positioning Accuracy on the Nanometer Scale and Illumination Spot Size that can be Dialed Down to Micrometer Scale



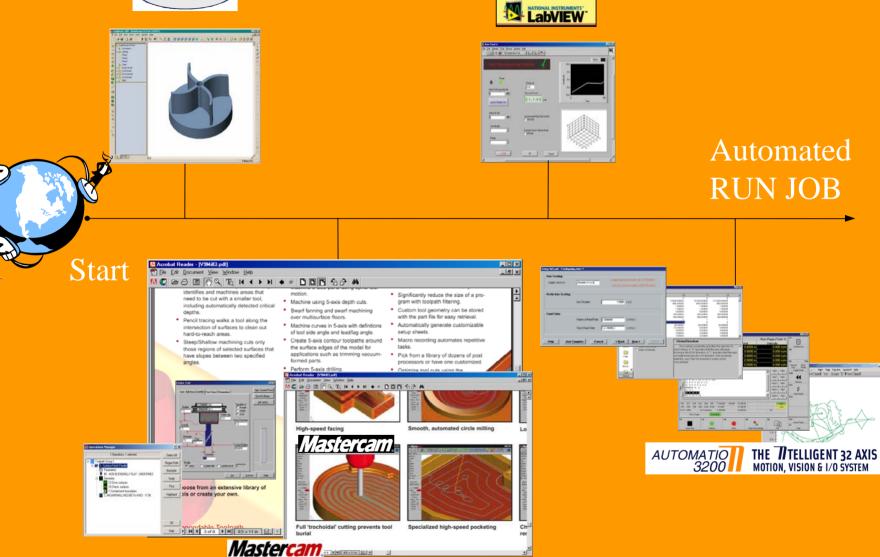


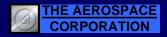
# Useful Attributes of the LMES



User Capabilities Microengineering & Rapid Prototyping Software Module Sequences







## Dynamic Control of Laser Power to Compensate for the Expected Changes in Motion Velocity that Occurs During Patterning



## High Fidelity Control of Laser Pulse Energy to "target"

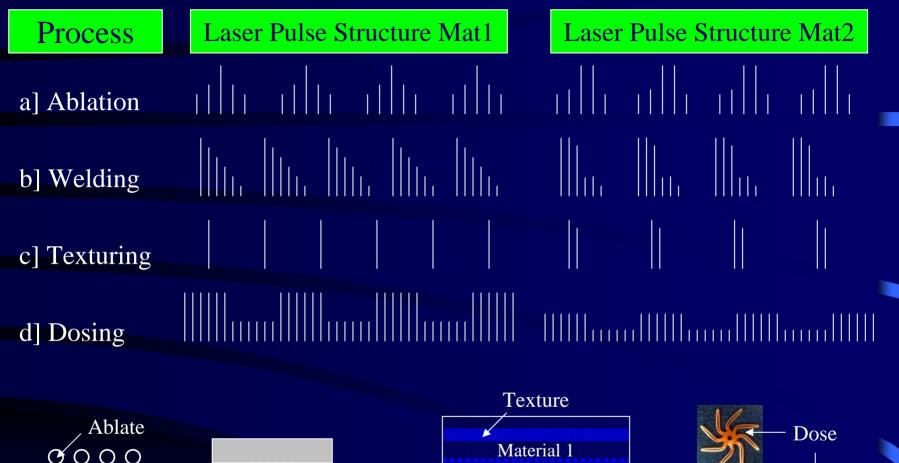
Laser or Target Motion

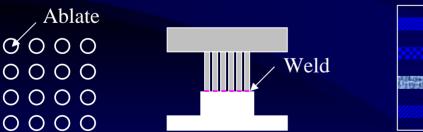
- I want to "script", a priori, the photon delivery (power, number of shots applied, types of lasers (color, pulse structure) etc.) for every laser SPOT SIZE along the tool path based on the *TYPE of Material Process that is to be Conducted*.
- Without regard to the motion speed/direction (i.e. vector velocity) and regardless of the feature pattern.
- We assume that the sample substrate properties have been mapped in 3D, a priori.
- We assume that the material has "properties" that can be expressed via controlled deposition of light.

Laser Spot

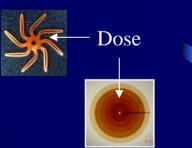
Size

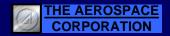
#### Laser Material Processing Using Modulated Pulse Sequences



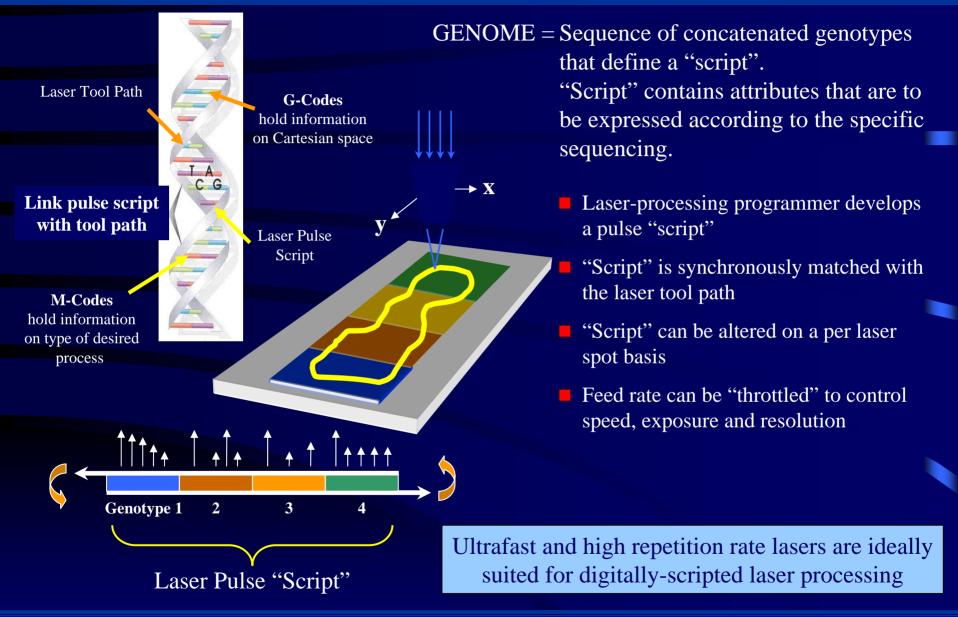






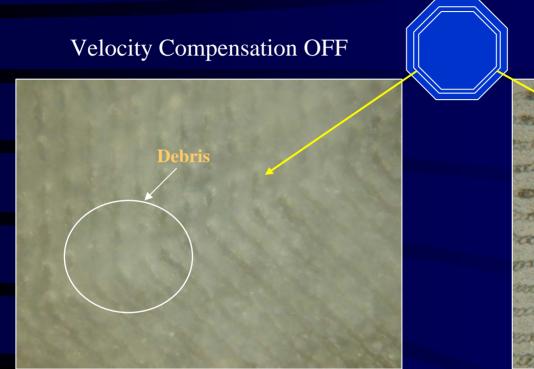


#### Use of Digitally-Scripted Genotype Pulse Patterns



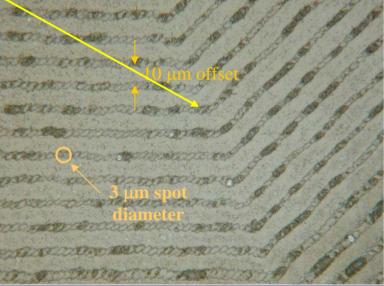


Ablation of Glass Using Modulated fs Laser Pulses at 400 nm



Spallation and debris formationFractures and thermal-induced stress

#### Velocity Compensation ON



Localized material removalPatterns are clean and well defined



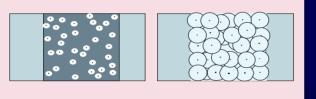


Processing Photoceramic Glasses Typical Process Flow

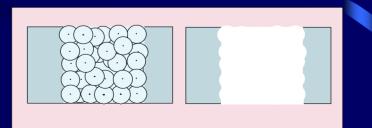
Step 1: UV Illumination/Latent Image Ce <sup>3+</sup> + hv (312nm, 2 J/cm<sup>2</sup>)  $\rightarrow$  Ce <sup>4+</sup> + e<sup>-</sup>

$$e^{-} + Ag^{+} \rightarrow Ag^{0}$$

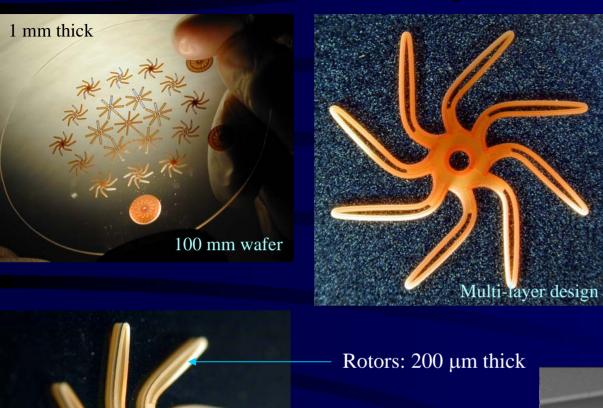
**Step 2: Ceramization to a Meta-Silicate** 



Step 3: Preferential Isotropic Etching
Crystalline Li<sub>2</sub>SiO<sub>3</sub> dissolves 20x faster than the amorphous glass in 5% hydrofluoric acid.
Li<sub>2</sub>SiO<sub>3</sub> + 3HF -> 2LiF + H<sub>2</sub>SiF<sub>6</sub> + 3 H<sub>2</sub>O

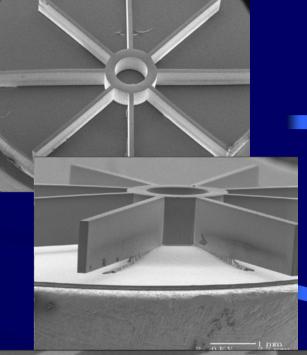


#### **Arrays of Meso-scopic Scale Microturbines**



Spindle depth: 500 µm

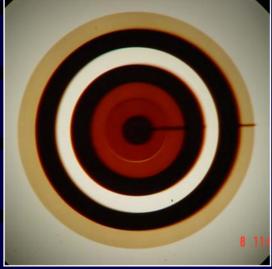
Trough: 300 µm



lmm

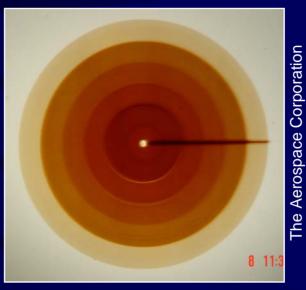
155µm

#### **Exposed and baked**

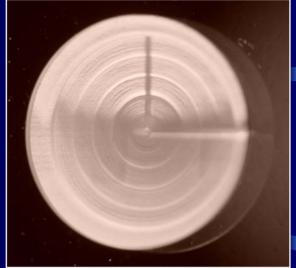


The Aerospace Corporation

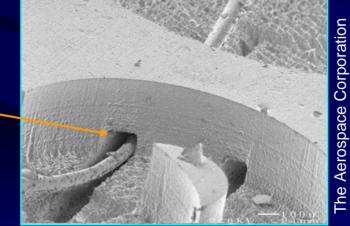
#### **Control of Optical Transmission** (IR)



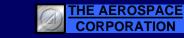
What time is it? **THz Fresnel Lens** 



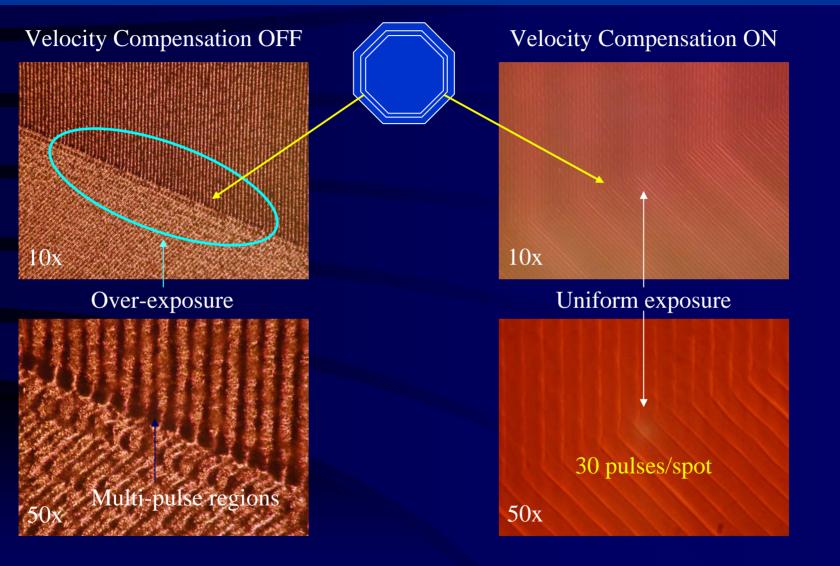
The Aerospace Corporation



#### Undercut **Structures**



Exposure of Photosensitive Glass Using Modulated fs Laser Pulses at 400 nm





Why is the The Aerospace Corporation, a Federally Funded Research & Development Center of Department of Defense Interested in the Development of a LMES?

Development of a Direct-Digital Manufacturing Methodology for the Design, Fabrication and Assembly of Small Satellites

Satellites that are Mass Producible and can be Mass-Customized

**Key: Materials in Satellite Must Serve a Multifunctional Role** 

Consider Glass-Ceramic Materials as Structural Support For Small Satellites

## The Co-Orbiting Satellite Assistant (COSA) Mission

**Differential GPS** 

Antenna for Comm-link to Host Vehicle and Earth

Micro-Platform (~ 100mm in diameter); contains GPS receiver, CCD camera, command receiver, data transmitter, micropropulsion, micro GN&C, C&DH, and primary battery

Thruster Plume

COSA

**Host Satellite** 

Co-Orbit at 700 km altitude: 99 minute Orbit Air drag makeup:  $\Delta V \sim 7m/s$ Observation Orbit:  $\Delta V \sim .180$  m/s Total Mission:  $\Delta V \sim 8m/s$ 

## **COSA Observation Trajectories**

#### **Co-orbital** with Inclination:

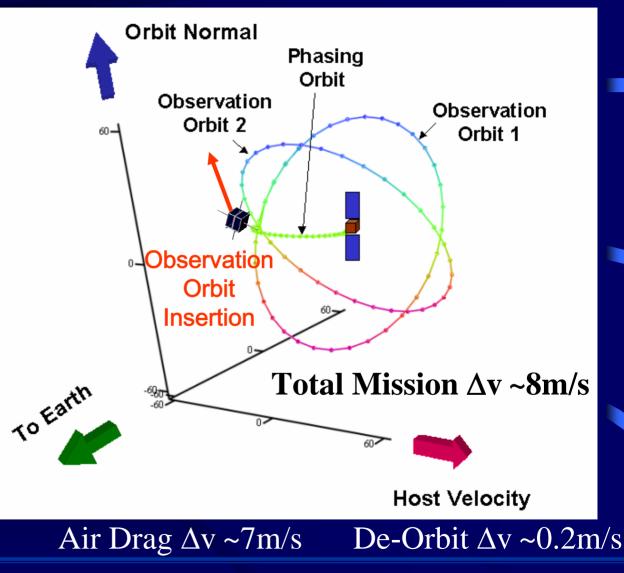
#### **Phasing orbit:**

- 1st impulse: 3 mm/s
- 2nd impulse: 3 mm/s
- 99 minutes @ 700 km
- mN thrust levels

#### **Observation orbits:**

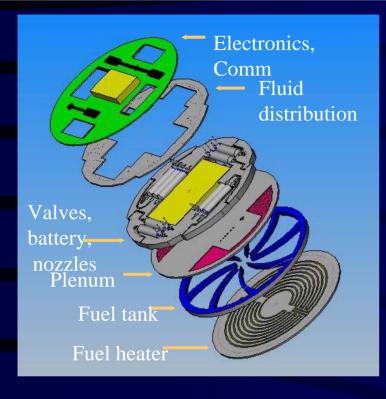
- Total: ~.180 m/s
- mN thrust levels

 $F\delta t = m \Delta v$  $\Delta v = 1mN(1s)/0.1Kg$  $\Delta v = 0.01m/sec$ 





#### **COSA: All Direct-Digital-Manufacturing**

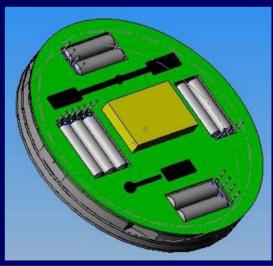


#### **Manufacturing Approach:**

- •Generate solid models of all patterned PSGC wafers and components.
  - Conduct tests for form fit,
  - Conduct mission analysis tests.
- Generate tool-path files for all laser direct-write operations.
- Material to be removed, deposited, altered.
  Translate tool-path code to patterning motion.
  Pattern via laser direct-write variable exposure

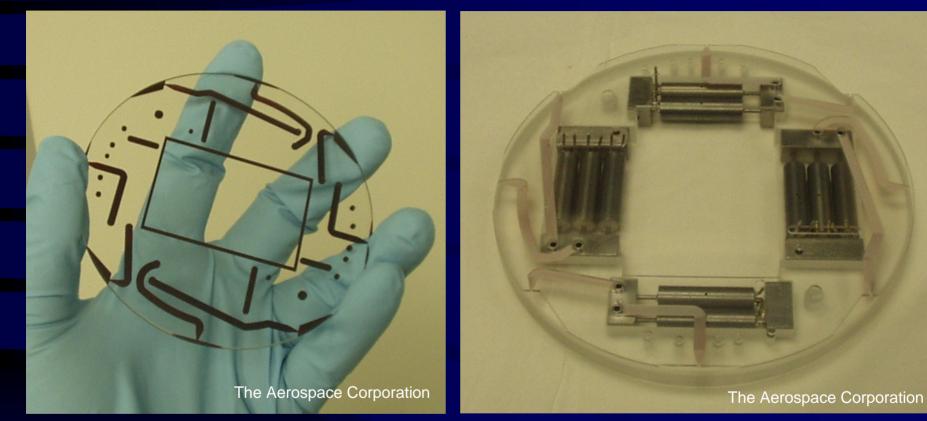
processing

A Propulsion System with Guidance, Navigation and Control (GNC) in PSGC Material





## **COSA Wafers**

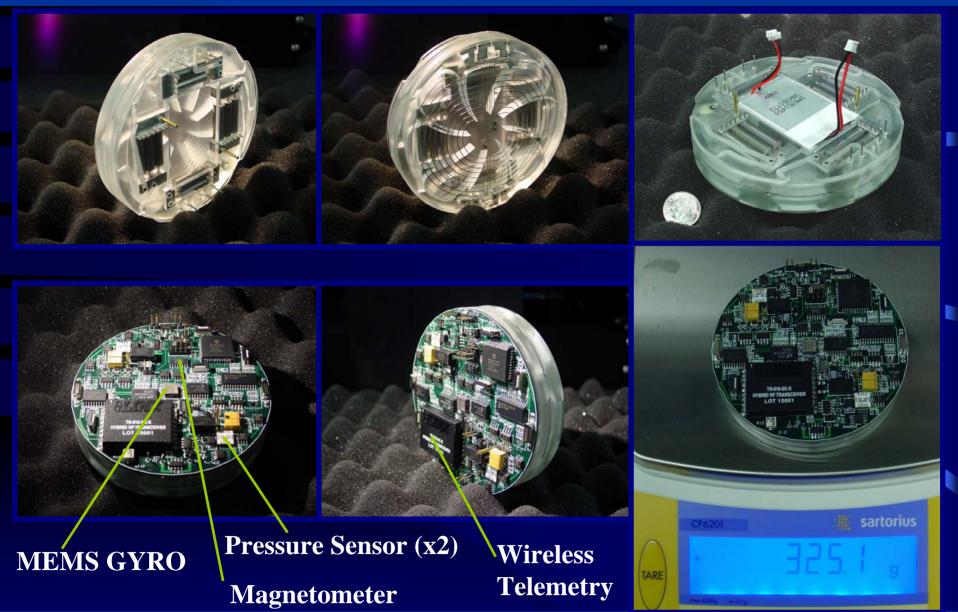


Exposed and baked nozzle layer

Etched nozzle and valve layers with valves



#### **"COSA: The Prototype Propulsion Module"**



Henry.Helvajian@aero.org Department of MicroNanotechnology

\*S. Janson, A. Huang, W. Hansen and H. Helvajian, J. AIAA 2004-6701



## **Orientation Control Tests for the COSA Vehicle "Hope"**

**Rotation on Air Table <u>Without</u> Thrust or Orientation Control** 



## IMU guided Rotational Control <u>with</u> pulsed thrusting





## **COSA Manufacturing Process: Prototype Vehicle**

COSA Process Time (hrs.)				
	Patterning	Baking	Polishing	Etching
Is Process Automated? Can it Run 24/7	Yes	yes	yes	yes
Wafer				
Heater	10	4	12	2
Fuel	0.25	0	0	0
Plenum	19	4	12	3
Spacer	2	4	12	1
Battery	0.25	0	0	0
Nozzle	17	4	12	3
Capping	7	4	12	1
SUM	55.5	20	60	10
Total Time to Fabricate COSA Wafers (Hrs)			145.5	
Total Time to Bond All Wafers (Hrs)			2	
Total Time to Bake Bonded Wafers (Hrs)			24	
Estimated Total Time for Electronics wafer assembly			36	
Total Time for COSA Vehicle Manufacturing (Hrs)			207.5	

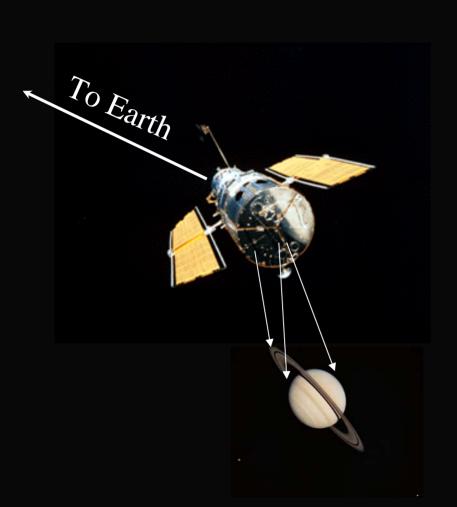


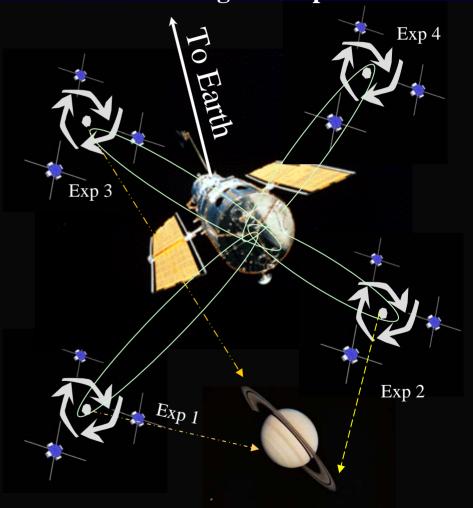
#### The LMES and the JLAB FEL can Serve as a Manufacturing Testbed for a High Throughput Patterning of Glass Ceramic Nanosatellite/COSA Parts

- A complete set of COSA vehicle wafers can be patterned in less than 30 minutes (instead of 55 hours).
- <u>Multiple</u> COSA vehicles can be assembled in batch mode in less than 30 hours (instead of 1 per 210 hrs).
- All digital direct manufacturing allows for design alterations to be done on the digital model of the vehicle which is directly realized in the processed part.



Changing the Approach on Future Space MissionsAdaptable Data Acquisition: Inspired by 3rdGeneration (3G) Cell Phone TechnologyCurrent Architecture DesignFuture Architecture DesignOne Spacecraft"Mother & Daughter" Spacecraft









## Conclusions

- Laser material processing is a growing industry world wide.
- Laser microengineering with the JLAB FEL can make high fidelity precision processes (e.g non thermally governed processes) commercially feasible for industrial use.
- A unique laser direct-write patterning tool (LMES) has been developed and installed at JLAB that permits the high fidelity controlled delivery of laser pulses based on a preprogrammed "script".
- Laser microenginering becomes <u>most</u> cost effective when the laser is used only to "activate" the material not to induce the desired physical transformation process.
- The JLAB FEL with LMES patterning tool is a test bed for the manufacturing of miniature glass/ceramic space systems by a Direct-Digital Development methodology. "radically alter small satellite design and manufacture paradigms" producing a nanosatellite space vehicle a day.



# Thank You